

National Aeronautics and Space Administration

Dynamic Radioisotope Power Systems: Convertor Development at NASA GRC

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2018 Power Community TDT/CLT

Dynamic Conversion Power System Background

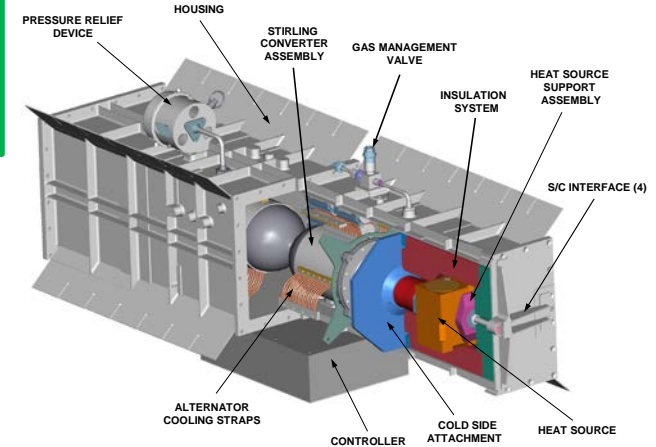


Advantages:

- Higher efficiency, less waste heat for spacecraft
- Low generator power decline (fuel decay only)
- Large multi-mission generator design space
- Extensible to high power levels

SRG-110 (2001 – 2006)

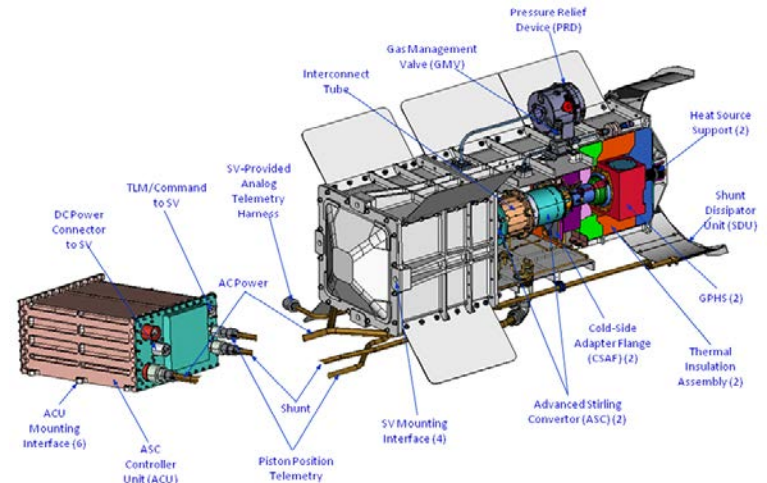
- ~114 W_e output
- Infinia's Technology Demonstration Convertor (TDC)
- 2 GPHS modules
- Overall efficiency = 23%
- 4.2 W_e/kg (before engineering unit build)



SRG110 (Lockheed Martin)

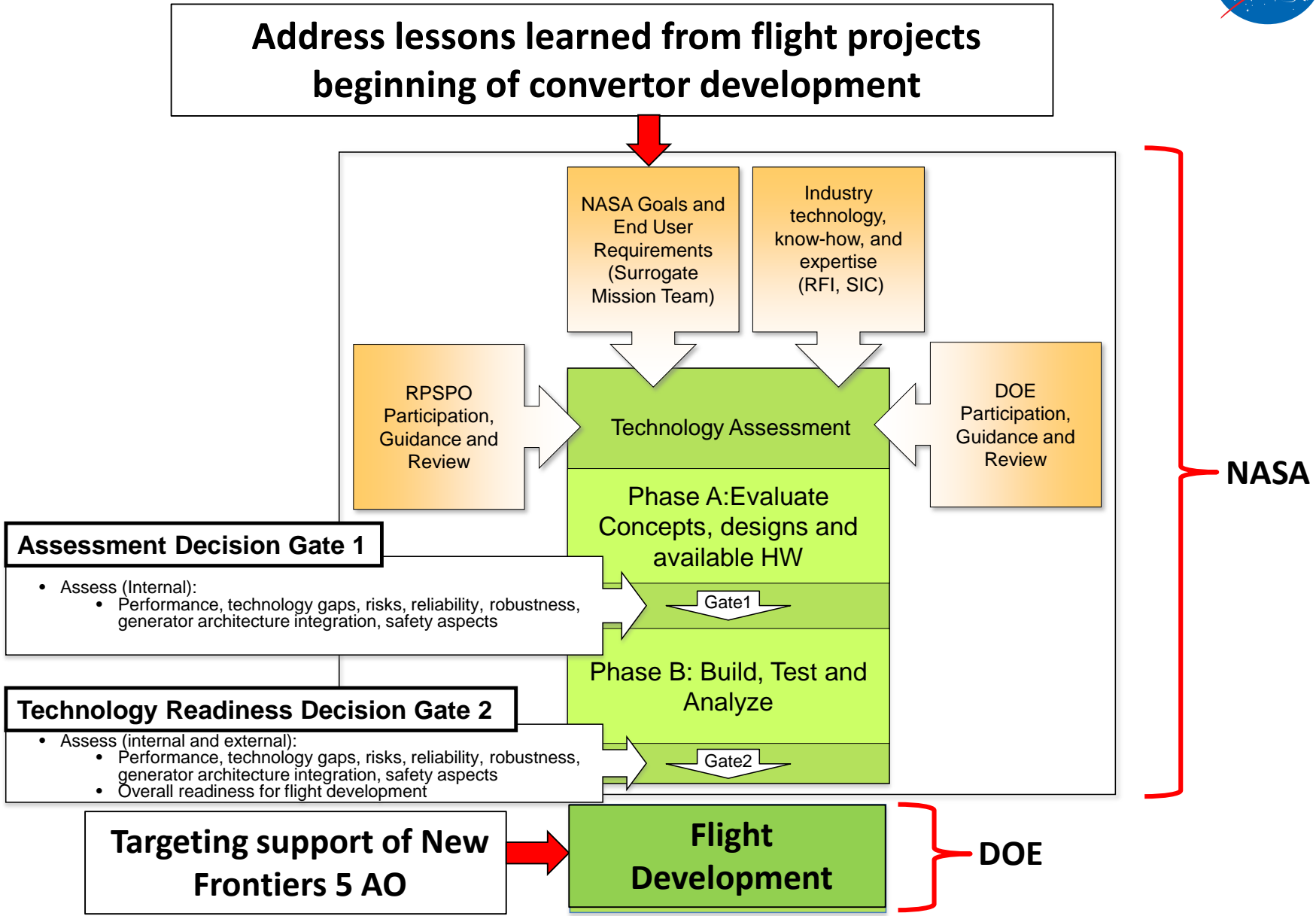
ASRG (2007 – 2013)

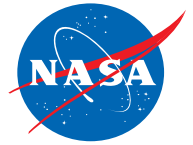
- ~140 W_e output
- Sunpower's Advanced Stirling Converter (ASC)
- 2 GPHS modules
- Overall efficiency = 28%
- 4.4 W_e/kg



ASRG (Lockheed Martin)

NASA's Dynamic Convertor Development Path



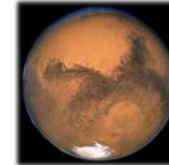


Key Converter Performance Goals

Item	Description
Life	20 years
Efficiency	$\geq 24\%$ at $T_{\text{cold}} > 100\text{ }^{\circ}\text{C}$
Specific Power	$20\text{ W}_e/\text{kg}$ (converter only)
Partial power	Can be throttled down to 50%
Degradation	$< 0.5\%$ / year
Hot-End Temp	$< 1000\text{ }^{\circ}\text{C}$
Cold-End Temp	20 to $175\text{ }^{\circ}\text{C}$
Random Vibe	Launch qual
Static Accel	20g for 1 minute, 5g for 5 days
Radiation	300 krad
Size	Enables generator that can fit in shipping container

Multi-Mission Capable:

Mars



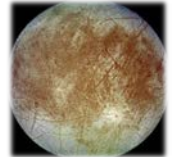
Titan



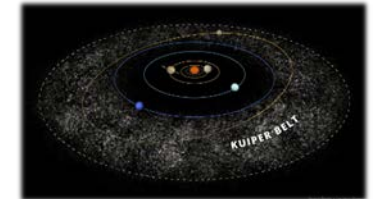
Moon



Europa



Deep Space



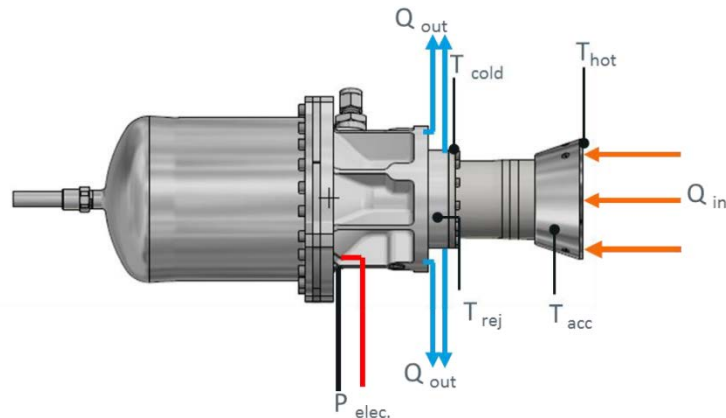
Robustness goals also defined:

- Design has margin to tolerate events outside expected environments
- Fewer single-point-failures is more robust
- Tolerant of loss of electrical load
- Tolerant of operational error
- Manufacturability not dependent on specialized workmanship

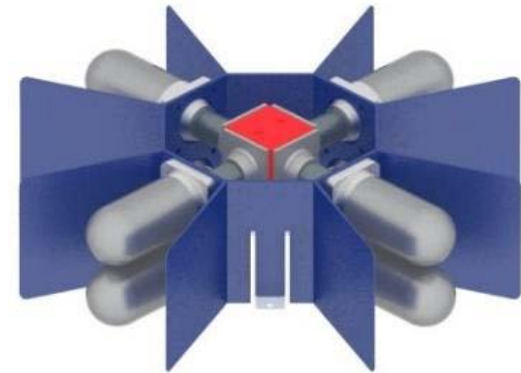
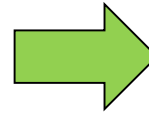


Flexure Isotope Stirling Convertor (FISC)

American SuperConductor (AMSC), formerly Infinia Tech Corp.



70 W Flexure Isotope Stirling Convertor (FISC)



Notional 240 W generator concept
with 100% convertor redundancy

Convertor Performance

Hot-end Temp	650 °C
Cold-end Temp	20 to 175 °C
Efficiency	31% @ $T_{COLD}=100^{\circ}\text{C}$
Power Output	70 W _{ac}
Mass	3.3 kg (>20W _e /kg)

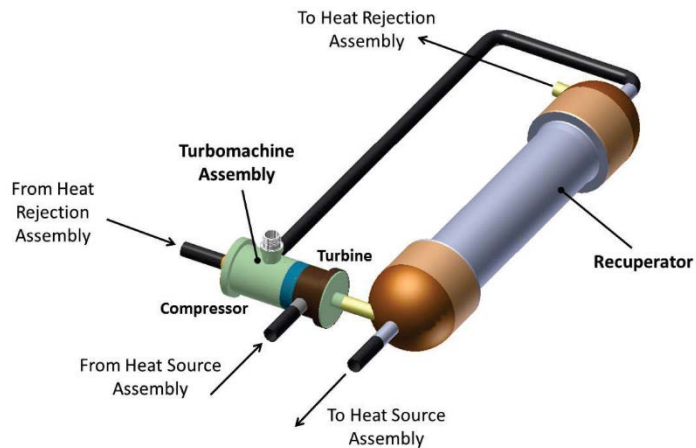
- Flexure-bearings, beta arrangement free-piston Stirling conv.
- Derivative of Technology Demonstration Convertor (TDC) from a 1990's SBIR and SRG-110 project
- Design deltas relative to TDC to improve the following:
 - Higher radial stiffness flexures, overstroke tolerance, hot-end temperature margin
 - Independently verifiable subassemblies
 - Higher efficiency alternator, higher cold-end temp capability
 - System integration : Tailored interfaces

Status:

**Decision Gate 1 successfully passed
Phase 2 in progress**

Turbo-Brayton Convertor (TBC)

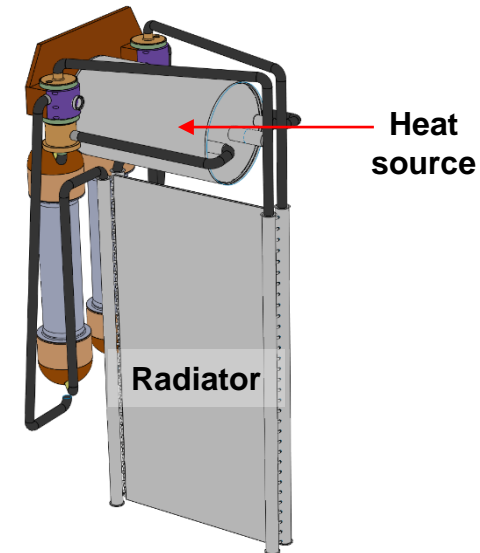
Creare, LLC



355 W_e Turbo-Brayton Convertor (TBC)

TBC Performance

Turbine Inlet Temp (Hot End)	730 °C
Compressor Inlet Temp (Cold End)	20 to 175 °C
Efficiency	26% @ T _{COLD} =100°C
Power Output	355 W _{ac}
Mass	15.5 kg (>20W _e /kg)



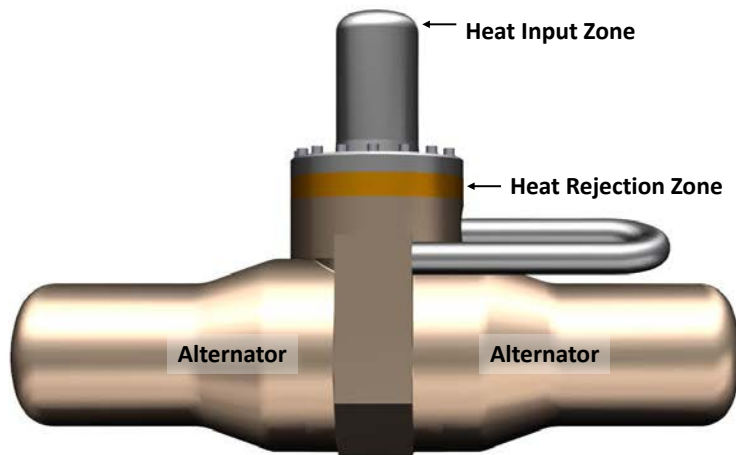
Notional 355 W_e generator concept with 100% convertor redundancy

- Closed Brayton continuous flow cycle with recuperation
- Scaled-down from previous designs
- Leverages heritage from Creare's HST NICMOS cooler
- Two counter-rotating units permits redundancy, and nullifies angular momentum

Status:
Decision Gate 1 successfully passed
Phase 2 in progress

Thermo-Acoustic Power Convertor (TAPC)

Northrop Grumman Aerospace Systems

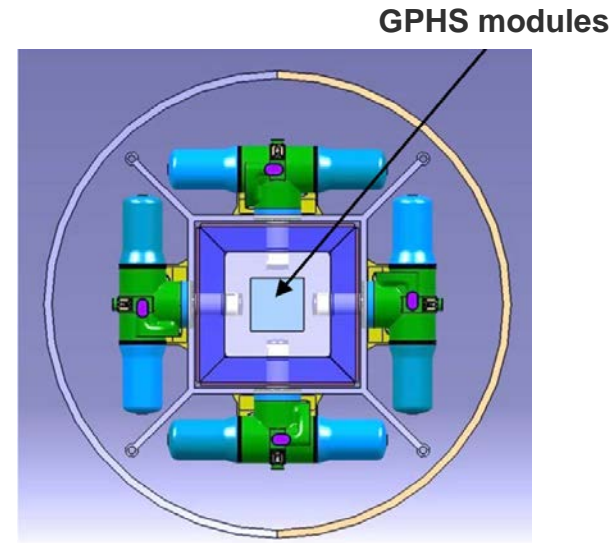
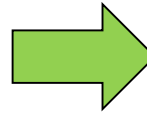


110 W_e Thermo-Acoustic Power Convertor
(TAPC)

TAPC Performance

Hot-end Temp	700°C
Cold-end Temp	20 to 100 °C
Efficiency	25% @ T _{COLD} =100°C
Power Output	110 W _{ac}
Mass	6.5 kg (< 20 W _e /kg)*

*Options being explored to reduce convertor mass to meet W/kg target



Notional 220 W_e generator concept with 100% convertor redundancy

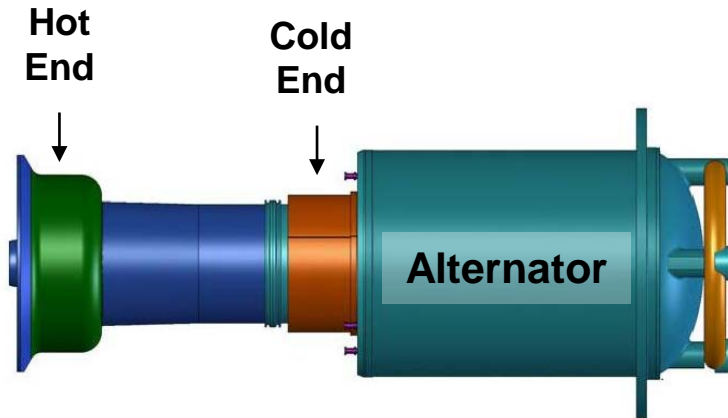
- Thermoacoustic Stirling cycle
- Eliminates physical displacer (no moving parts in hot end)
- Natively balanced, dual-opposed alternator building block
- Alternators driven by shared compression space
- Based on previous development efforts:
2003 SBIR, IRAD-developed device

Status:
Decision Gate 1 Design Review Completed
Phase 2 pending RFA responses

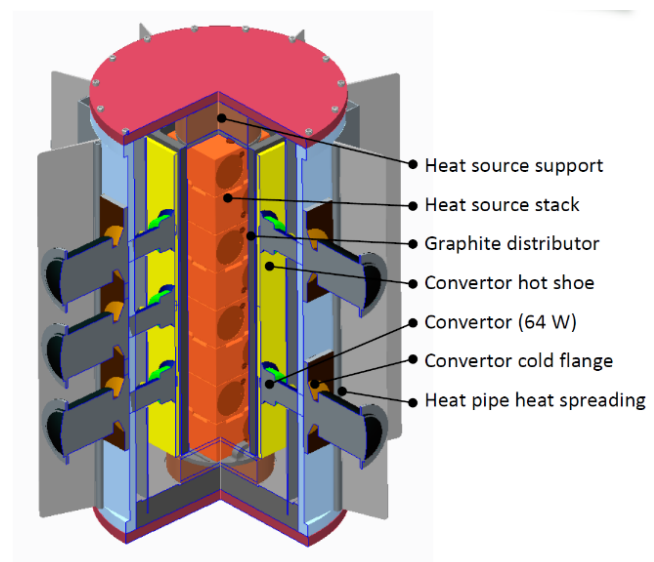
Sunpower Robust Stirling Convertor (SRSC)



Sunpower, Inc.



64 W_e Sunpower Robust Stirling Convertor (SRSC)



Notional 500 W_e generator concept with 25% convertor redundancy

SRSC Performance

Hot-end Temp	640°C
Cold-end Temp	20 to 175 °C
Efficiency	30% @ T _{COLD} =100°C
Power Output	64 W _{ac}
Mass	1.6 kg (> 20 W _e /kg)

- Gas-bearing based free-piston Stirling
- Derivative of Advanced Stirling Convertor (ASC) from ASRG Project
- Enables wide generator design space
- Design deltas relative to ASC to improve the following:
 - Higher radial gas bearing stiffness, overstroke tolerance, regenerator robustness, debris tolerance
 - Higher cold-end temp capability, static acceleration

Status:
Decision Gate 1 Design Review Completed
Phase 2 pending RFA responses



Path to Flight

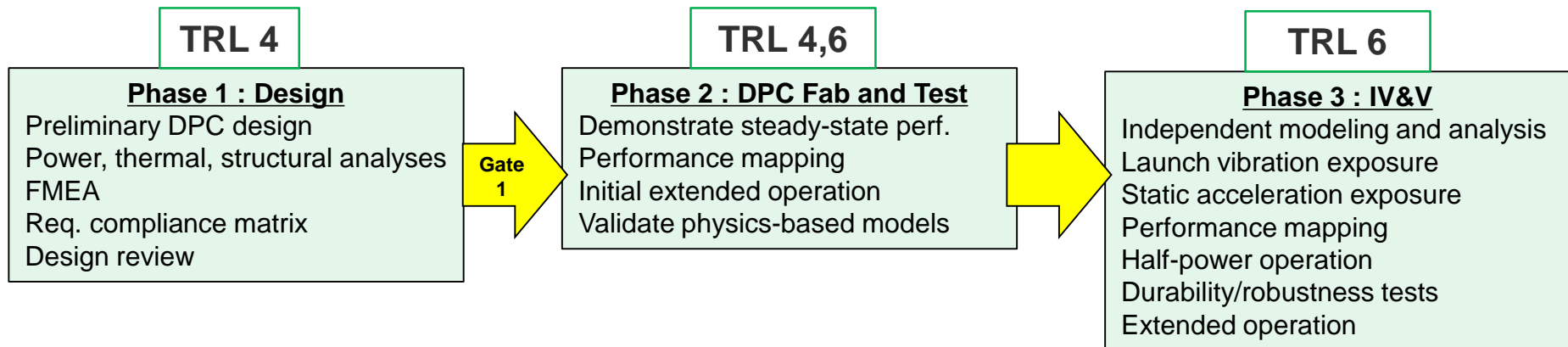
Goal:
Achieve convertor TRL 6, then initiate generator flight development

per NASA/SP-2007-6105 Rev1, pg 296:

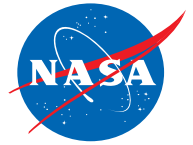
“System/subsystem model or prototype demonstration in a relevant environment (ground or space)”

Surrogate Mission Team (SMT), chartered by RPS Program

- NASA, DOE, JPL, APL, GSFC
- Formulated requirements to provide mission pull
- Integrated with DPC contract progress monitoring
- Formulated a TRL evaluation method
- Providing failure mode and probability of success analysis
- Work phases and deliverables tied to TRL advancement



Stirling Convertor Extended Operation

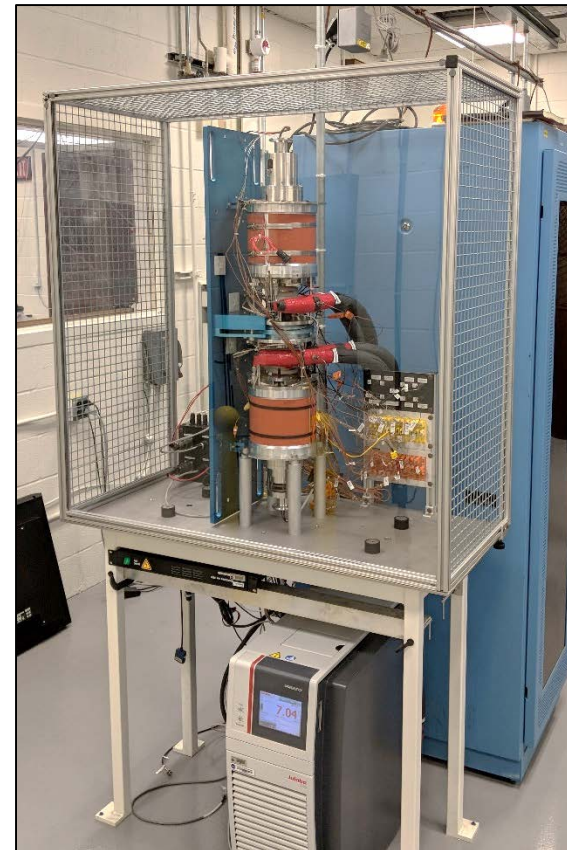


Many convertors from SRG110 and ASRG projects are still undergoing continuous operation today

Project & Provider	Test Article	Hrs of Operation
SRG 110 Infinia, Corp.	TDC #13	108,509
	TDC #15 & #16	100,155 each
	SES #2 (SRG-110 eng unit)	1,887
ASRG Sunpower, Inc.	ASC-E3 #4* , #9	26,431 / 12,472
	ASC-E3 #6* , #8	19,505 / 15,358
	ASC-0 #3	71,361
	ASC-L	33,223

Cumulative Per-Convertor Runtime as of Feb 2018

*Have undergone random vbe portion of life certification
(20 years = 175,000 hrs)



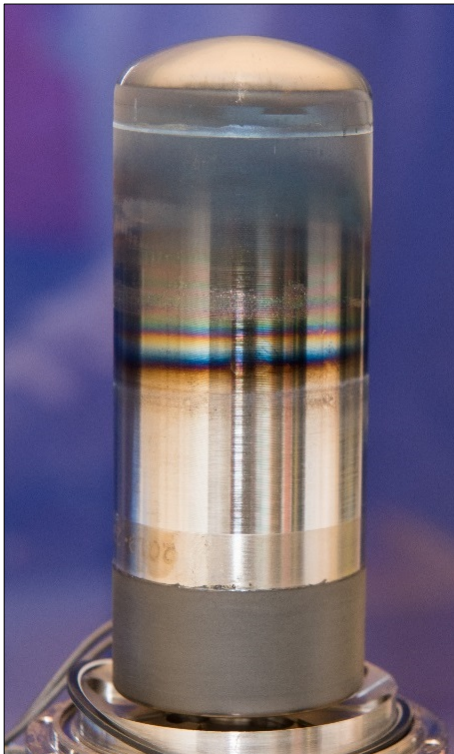
ASC-E3 Pair Extended Operation Test Article

TDC #14 Disassembly and Inspection

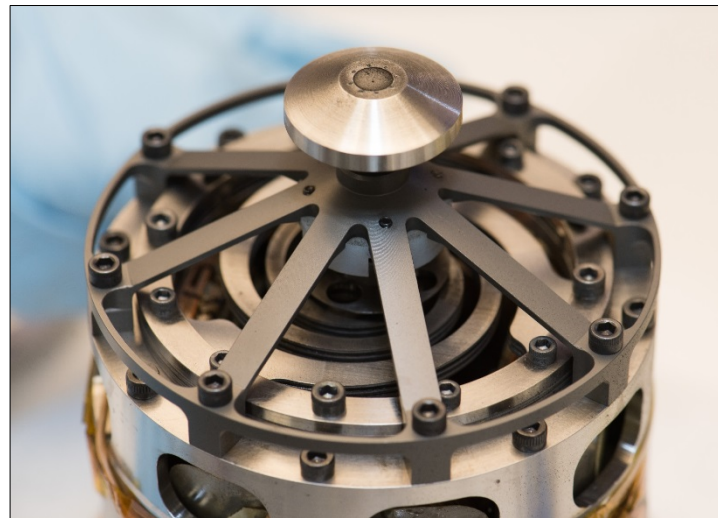


Encouraging results from TDC #14 inspection
105,620 hrs of operation = 12 years, 31 billion cycles
Further disassembly is planned

- No sign of flexure degradation
- Signs of oxidation on expected surfaces – likely from early non-hermetic operation
- Geometric stability verified via Coordinate Measuring Machine (CMM)
- Evidence of oxide residue/dust in various areas – did not degrade operation



**TDC #14 displacer after
12 years of operation**



**TDC #14 aft flexure stack after 12
years of operation**

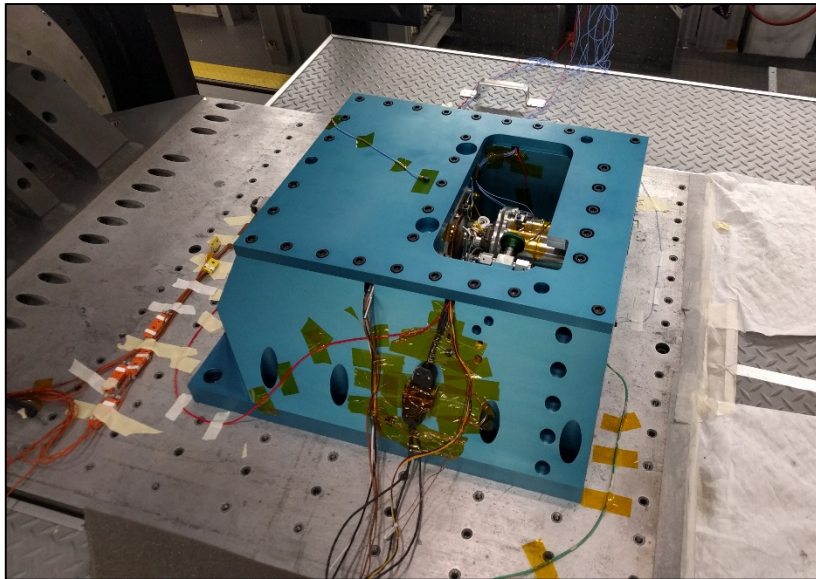


**TDC #14 piston after 12 years
of operation**

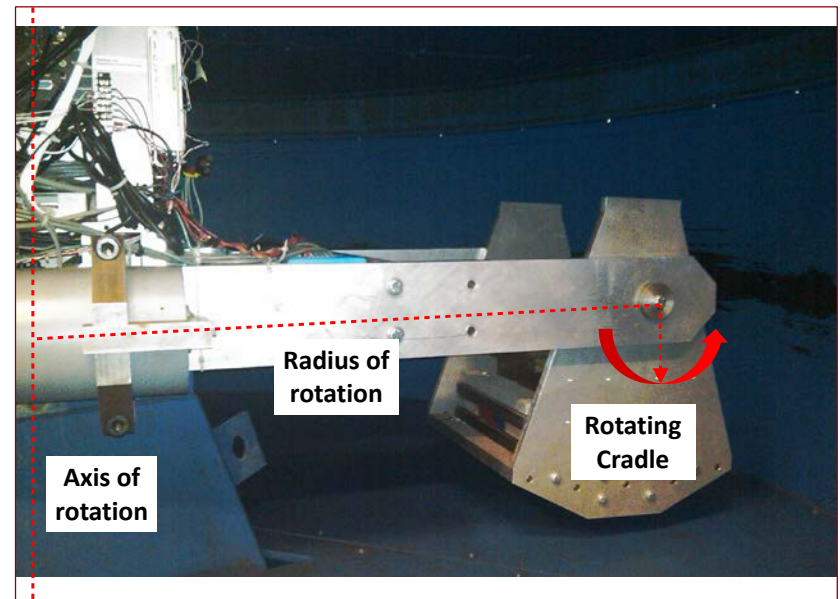
Launch Vibration Exposure on SES #2

Engineering Unit convertor from SRG-110 project successfully passed launch simulation while operating

- **10.35 g_{rms} profile formulated by SMT, encompasses wide span of launch vehicles**
- **2 min duration at full random vibe level:** Had to reduce piston amplitude for axial exposure (expected), temporary reduction in power output during lateral axes exposures (expected)
- **Static acceleration exposure up to 5g axial and 20g lateral successfully completed in April 2018**
- **SES #2 now operating continuously at full power, 1800 hrs accumulated**

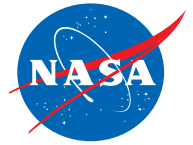


SES #2 undergoing launch vibration exposure



Centrifuge facility for static acceleration tests
(Case Western Reserve University)

Conclusions



NASA's dynamic power convertor development in support of high-efficiency RPS is progressing as planned, and shows promise

- **2 DPC contracts have passed Decision Gate 1, and Phase 2 has started (convertor prototype fabrication and test)**
- **2 DPC contracts have completed Phase 1 design reviews, RFAs being worked now to enable decision for Phase 2**
- **NASA GRC is preparing for DPC prototype IV&V, ~2020**
- **Ongoing research utilizing existing hardware supports viability of dynamic power conversion for RPS**